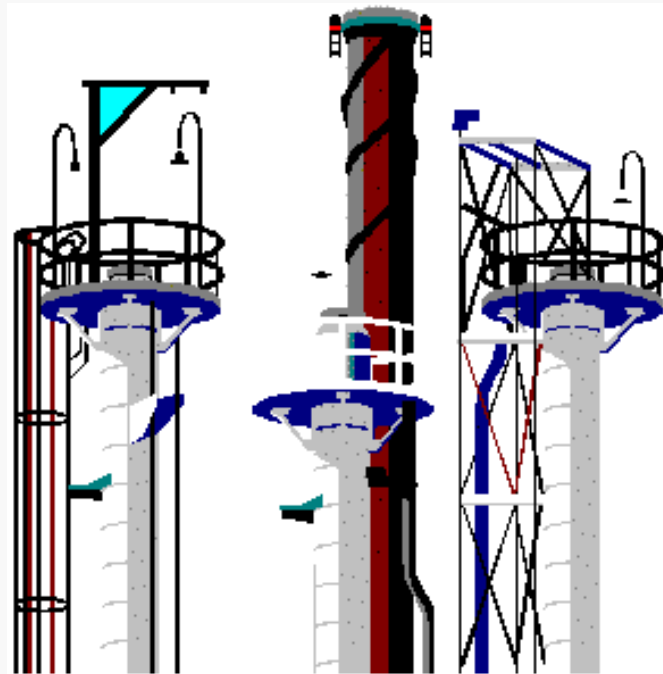


Ch2 Principles of Momentum Transfer and Overall Balance



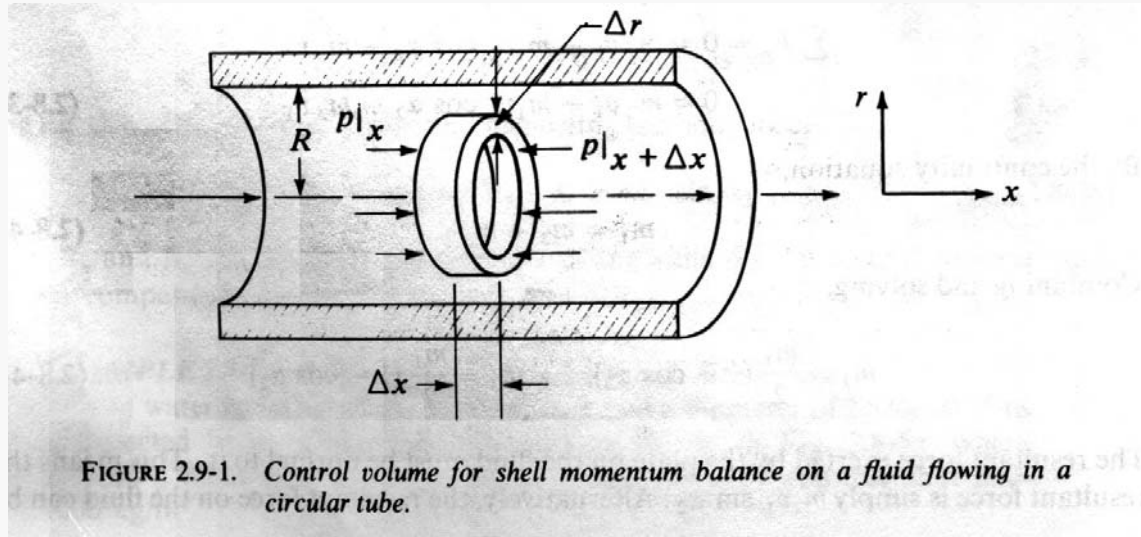
이 광남

운동량 전달 원리 및 총괄수지

2.9 Shell Momentum Balance and velocity profiles in laminar Flow

- ▶ 셸 운동량 수지
 - ~ 셸(미소 대상부피)에 대해 운동량 수지 적용
 - 셸을 미분 크기로 줄여주고
 - 점도 법칙을 적용
 - 속도분포, 압력에 관한 식 유도
 - ↳ ▶ 평균속도, 최대속도, 전단응력

2.9B Shell Momentum Balance inside a Pipe



- ▶ Assumption
 1. incompressible, Newtonian fluid
 2. one-dimensional, steady state, laminar flow
 3. fully developed flow
 4. no-slip condition

$$\left\{ \begin{array}{l} \text{rate of accumulation} \\ \text{of momentum} \\ \text{in system} \end{array} \right\} = \left\{ \begin{array}{l} \text{rate of} \\ \text{momentum in} \end{array} \right\} - \left\{ \begin{array}{l} \text{rate of} \\ \text{momentum out} \end{array} \right\} + \left\{ \begin{array}{l} \text{sum of forces} \\ \text{acting on system} \end{array} \right\}$$

$$\begin{aligned} & \rho v_x (2\pi r \Delta r) v_x \Big|_x - \rho v_x (2\pi r \Delta r) v_x \Big|_{x+\Delta x} \\ & + 2\pi r \Delta x (\tau_{rx}) \Big|_r - 2\pi r \Delta x (\tau_{rx}) \Big|_{r+\Delta r} \\ & + P(2\pi r \Delta r) \Big|_x - P(2\pi r \Delta r) \Big|_{x+\Delta x} = 0 \end{aligned}$$

$2\pi r\Delta r\Delta x$ 로 나누고

$$\frac{(r\tau_{rx})|_r - (r\tau_{rx})|_{r+\Delta r}}{r\Delta r} + \frac{P|_x - P|_{x+\Delta x}}{\Delta x} = 0$$

*fully developed flow*내에서는 압력구배 일정

$$\frac{P|_x - P|_{x+\Delta x}}{\Delta x} = \frac{\Delta P}{\Delta x} \rightarrow \frac{\Delta P}{L} (= \frac{P_0 - P_L}{L})$$

$$\Delta r \rightarrow 0$$

$$\lim_{\Delta r \rightarrow 0} \frac{(r\tau_{rx})|_{r+\Delta r} - (r\tau_{rx})|_r}{\Delta r} = \left(\frac{\Delta P}{L}\right)r$$

$$\frac{d(r\tau_{rx})}{dr} = \left(\frac{\Delta P}{L}\right)r$$

$$\frac{d(r\tau_{rx})}{dr} = \left(\frac{\Delta P}{L}\right)r$$

$$r\tau_{rx} = \left(\frac{\Delta P}{L}\right)\frac{r^2}{2} + C_1$$

$$\tau_{rx} = \left(\frac{\Delta P}{L}\right)\frac{r}{2} + \frac{C_1}{r}$$

$$B.C.1 \text{ at } r=0, \tau_{rx} \neq \infty \rightarrow C_1 = 0$$

$$\tau_{rx} = \left(\frac{\Delta P}{L}\right)\frac{r}{2} = \left(\frac{\Delta P}{2L}\right)r$$

*momentum flux*는 r 에 대해 선형적으로 증가
 $r = R$ 에서 최대값

$$\tau_{rx} = \left(\frac{\Delta P}{L}\right) \frac{r}{2} = \left(\frac{\Delta P}{2L}\right)r = \left(\frac{P_0 - P_L}{2L}\right)r$$

$$-\mu \frac{dv_x}{dr} = \left(\frac{P_0 - P_L}{2L}\right)r$$

$$\frac{dv_x}{dr} = -\left(\frac{P_0 - P_L}{2\mu L}\right)r$$

$$v_x = -\left(\frac{P_0 - P_L}{4\mu L}\right)r^2 + C_2$$

B.C.2 at $r = R$, $v_x = 0$ (no slip condition)

$$C_2 = \left(\frac{P_0 - P_L}{4\mu L}\right)R^2$$

$$v_x = \left(\frac{P_0 - P_L}{4\mu L}\right)R^2 \left[1 - \left(\frac{r}{R}\right)^2\right] \quad \textit{parabolic velocity distribution}$$

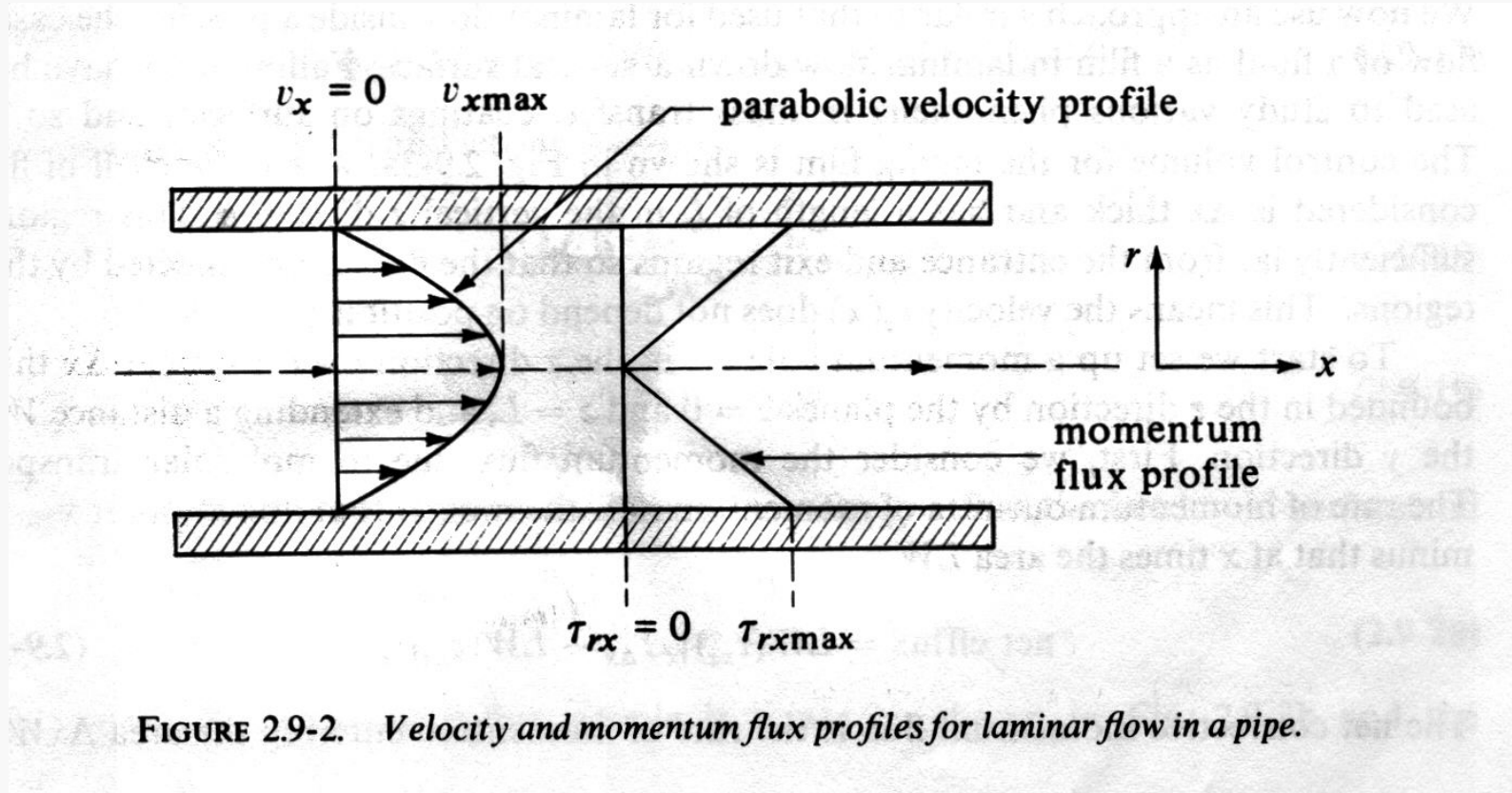


FIGURE 2.9-2. Velocity and momentum flux profiles for laminar flow in a pipe.

$$v_x = \left(\frac{P_0 - P_L}{4\mu L}\right)R^2\left[1 - \left(\frac{r}{R}\right)^2\right]$$

maximum velocity

at $r = 0$, $v_x = v_{x\max}$

$$v_{x\max} = \left(\frac{P_0 - P_L}{4\mu L}\right)R^2$$

$$v_x = v_{x\max}\left[1 - \left(\frac{r}{R}\right)^2\right]$$

average velocity

$$\langle v_x \rangle = \frac{1}{A} \iint_A v_x dA = \frac{v_{x\max}}{2} = \frac{(P_0 - P_L)R^2}{8\mu L} = \frac{(P_0 - P_L)D^2}{32\mu L}$$

; Hagen – Poiseuille eqn

$$\dot{Q} = \langle v_x \rangle A = \frac{\pi(P_0 - P_L)R^4}{8\mu L}$$

x-component of the forces of the fluid
on the wetted surface of pipe

$$\begin{aligned}
 -R_x &= (2\pi RL)(\tau_{rx})\Big|_{r=R} \\
 &= (2\pi RL)\left(\frac{P_0 - P_L}{2L} R\right) = \pi R^2 (P_0 - P_L) = \frac{32\mu L \dot{Q}}{D^2} \\
 \dot{Q} &= \frac{\pi(P_0 - P_L)R^4}{8\mu L} \rightarrow \pi R^2 (P_0 - P_L) = \frac{8\mu L \dot{Q}}{R^2} = \frac{32\mu L \dot{Q}}{D^2}
 \end{aligned}$$

2.9C Shell Momentum Balance for Falling Film

Shell~
길이 L
폭 W
두께 Δx

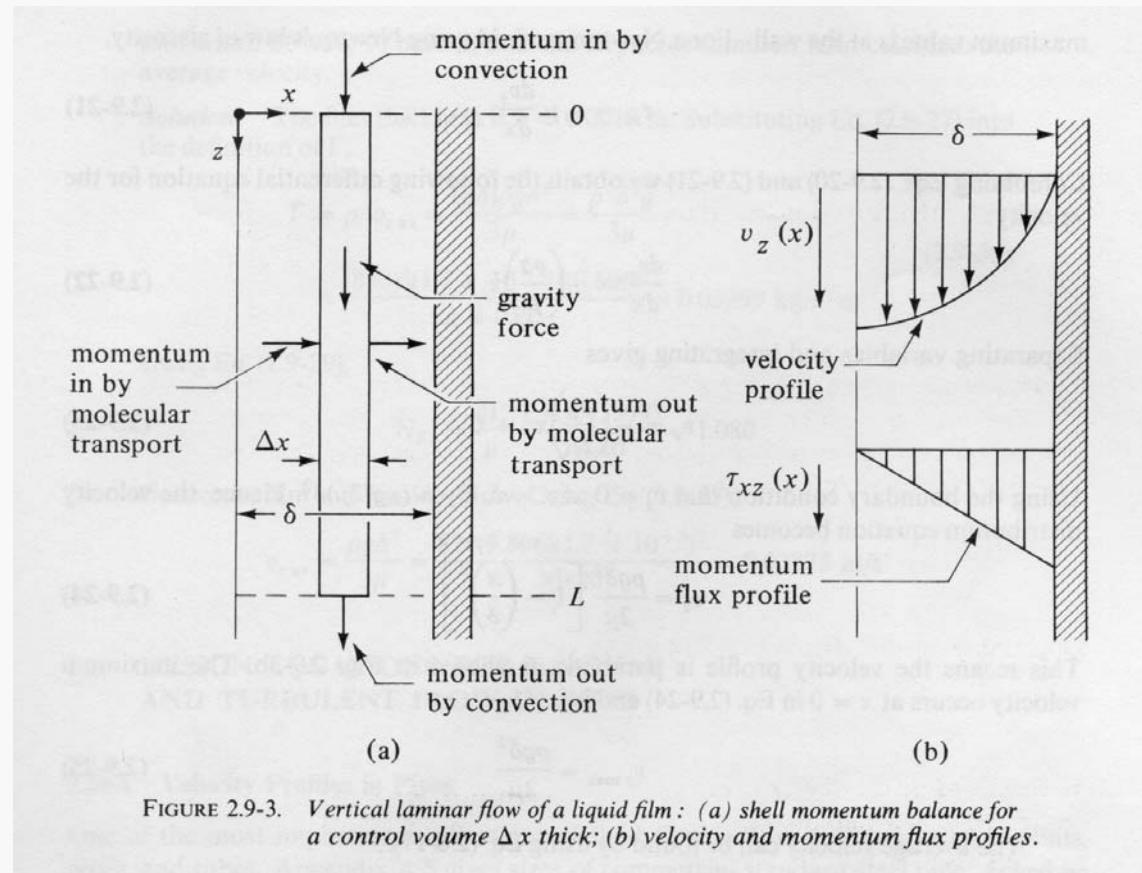


FIGURE 2.9-3. Vertical laminar flow of a liquid film : (a) shell momentum balance for a control volume Δx thick; (b) velocity and momentum flux profiles.

Assumption :

$$\left\{ \begin{array}{l} \text{rate of accumulation} \\ \text{of momentum} \\ \text{in system} \end{array} \right\} = \left\{ \begin{array}{l} \text{rate of} \\ \text{momentum in} \end{array} \right\} - \left\{ \begin{array}{l} \text{rate of} \\ \text{momentum out} \end{array} \right\} + \left\{ \begin{array}{l} \text{sum of forces} \\ \text{acting on system} \end{array} \right\}$$

$$\rho v_z (W\Delta x) v_z \Big|_z - \rho v_z (W\Delta x) v_z \Big|_{z+\Delta z}$$

$$+ WL(\tau_{xz}) \Big|_x - WL(\tau_{xz}) \Big|_{x+\Delta x}$$

$$+ WL\Delta x(\rho g) = 0$$

$WL\Delta x$ 로 나누고 $\Delta x \rightarrow 0$

$$\lim_{\Delta x \rightarrow 0} \frac{(\tau_{xz}) \Big|_{x+\Delta x} - (\tau_{xz}) \Big|_x}{\Delta x} = \rho g$$

$$\frac{d\tau_{xz}}{dx} = \rho g$$

$$\frac{d\tau_{xz}}{dx} = \rho g \text{ 에서}$$

$$\tau_{xz} = \rho g x + C_1$$

$$B.C.1 \text{ at } x=0, \tau_{xz} = 0$$

$$\tau_{xz} = \rho g x \text{ momentum flux 분포는 선형}$$

see Fig2.9-3

$$-\mu \frac{dv_z}{dx} = \rho g x$$

$$\frac{dv_z}{dx} = -\frac{\rho g x}{\mu}$$

$$v_z = -\frac{\rho g x^2}{2\mu} + C_2$$

B.C.2 at $x = \delta$, $v_z = 0$ (no slip condition)

$$C_2 = \frac{\rho g \delta^2}{2\mu}$$

$$v_z = \frac{\rho g \delta^2}{2\mu} \left[1 - \left(\frac{x}{\delta} \right)^2 \right] \text{ parabolic velocity profile}$$

maximum velocity

at $x = 0$, $v_z = v_{z,max}$

$$v_{z,max} = \frac{\rho g d^2}{2\mu}$$

average velocity

$$\langle v_z \rangle = \frac{1}{A} \iint_A v_z dA = \frac{\int_0^W \int_0^d v_z dx dy}{\int_0^W \int_0^d dx dy} = \frac{W}{Wd} \int_0^d v_z dx = \frac{1}{d} \int_0^d v_{z,max} \left[1 - \left(\frac{x}{d} \right)^2 \right] dx$$

$$= \frac{1}{d} v_{z,max} \frac{2d}{3}$$

$$\langle v_z \rangle = \frac{2}{3} v_{z,max} = \frac{\rho g d^2}{3\mu}$$

volumetric flow rate

$$\dot{Q} = \langle v_z \rangle A = \langle v_z \rangle (Wd) = \frac{\rho g d^2}{3\mu} (Wd) = \frac{\rho g d^3 W}{3\mu}$$

mass flow per unit width (G)

$$G = \frac{\dot{m}}{W} = \frac{\rho \langle v_z \rangle A}{W} = \frac{\rho \langle v_z \rangle Wd}{W} = \rho \langle v_z \rangle d$$

$$Re = \frac{4G}{\mu}$$

$Re < 1200$ *Laminar flow*

z – component of the force of the fluid
on the surface

$$\begin{aligned} -R_z &= \int_0^L \int_0^W (\tau_{xz}|_{x=\delta}) dydz = \int_0^L \int_0^W (\rho g \delta) dydz \\ &= (LW\delta)\rho g (= mg) \end{aligned}$$

